



Motivation and Outline

- Auroral charging of spacecraft is an important class of space weather impacts on technological systems in low Earth orbit
- In order for space weather models to accurately specify auroral charging environments, they must provide the appropriate plasma environment characteristics responsible for charging
- Improvements in operational space weather prediction capabilities relevant to charging must be tested against charging observations

Outline

- Spacecraft charging physics
- DMSP auroral charging
- ISS solar array and auroral charging
- Characteristics of auroral charging environments
- Space environment impacts database

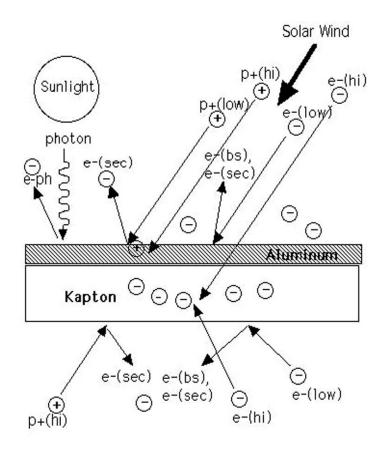
Acknowledgment: DMSP SSJ data provided by NOAA National Geophysics Data Center courtesy of the US Air Force



Surface Charging Physics

 Auroral charging is a process of balancing currents to and from spacecraft surfaces as a function of the spacecraft potential

$$\begin{split} \frac{dQ}{dt} = & C \frac{dV}{dt} = \frac{d\sigma}{dt} \, A = \sum_k I_k \\ \frac{dQ}{dt} = & \sum_k I_k = \\ & + I_i(V) \quad \text{incident ions} \\ & - I_e(V) \quad \text{incident electrons} \\ & + I_{bs,e}(V) \quad \text{backscattered electrons} \\ & \pm I_c(V) \quad \text{conduction currents} \\ & + I_{se}(V) \quad \text{secondary electrons due to } I_e \\ & + I_{si}(V) \quad \text{secondary electrons due to } I_i \\ & + I_{ph,e}(V) \quad \text{photoelectrons} \end{split}$$



(Garrett and Minow, 2004)



DMSP Charging





DMSP Auroral Charging

 Low energy (E₀ ~ 0) background ions accelerated by the spacecraft potential show up as sharp "line" of high ion flux in single channel

$$E = E_0 + q\Phi$$

- Assume initial energy E₀ = 0 with singly charge ions (O+, H+) and read potential (volts) directly from ion line energy (eV)
- DMSP SSJ4, SSJ5 detectors

Electrons: 20 channels

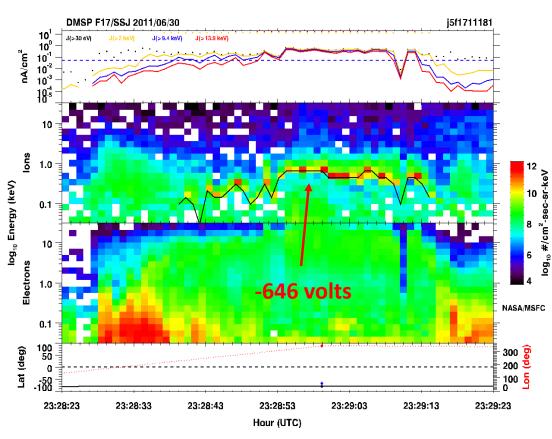
30 eV to 30 keV

lons: 20 channels

30 eV to 30 keV

 Nominal channel energies used for this work



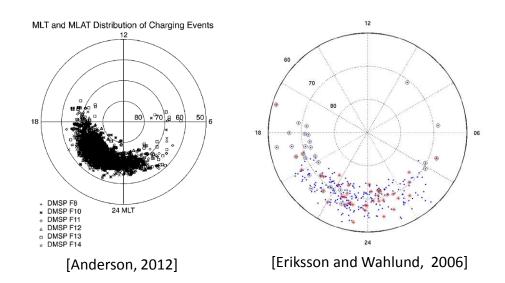




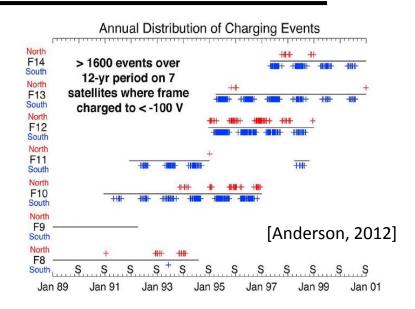
Auroral Charging Conditions

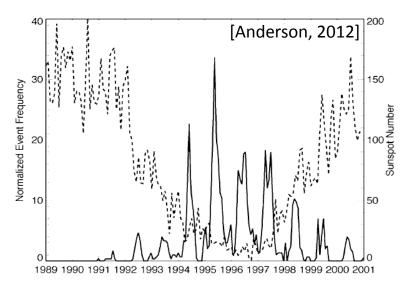
Necessary conditions for high-level (≥100 V) auroral charging*

- No sunlight (or ionosphere below spacecraft in darkness)
- Intense electron flux >10⁸ e/cm²-s-sr at energies of 10's keV
- Low ambient plasma density (<10⁴ #/cm³)



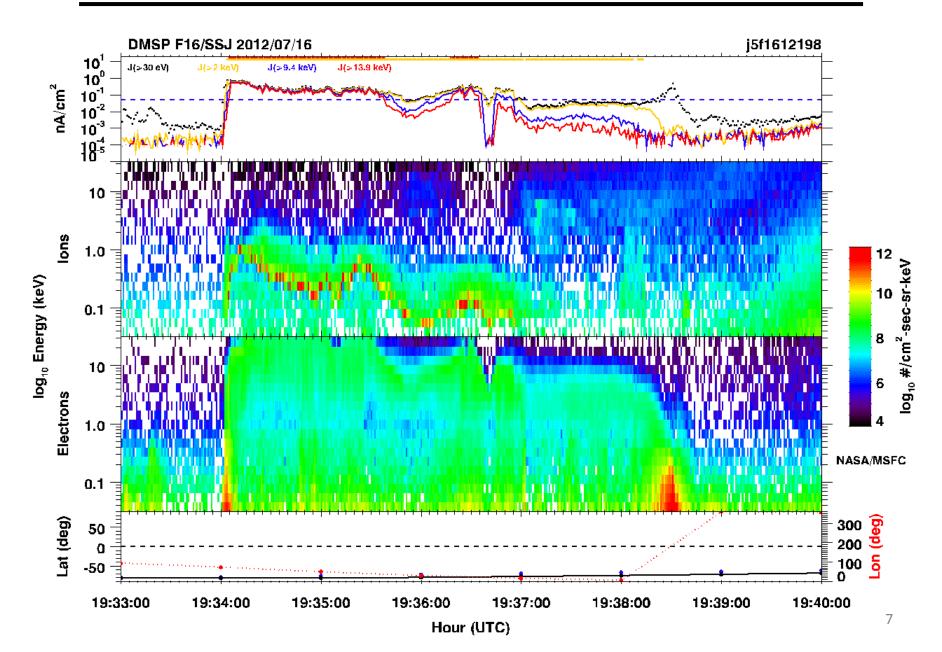
^{*}Gussenhoven et al., 1985; Frooninckx and Sojka, 1992; Eriksson and Wahlund, 2006.





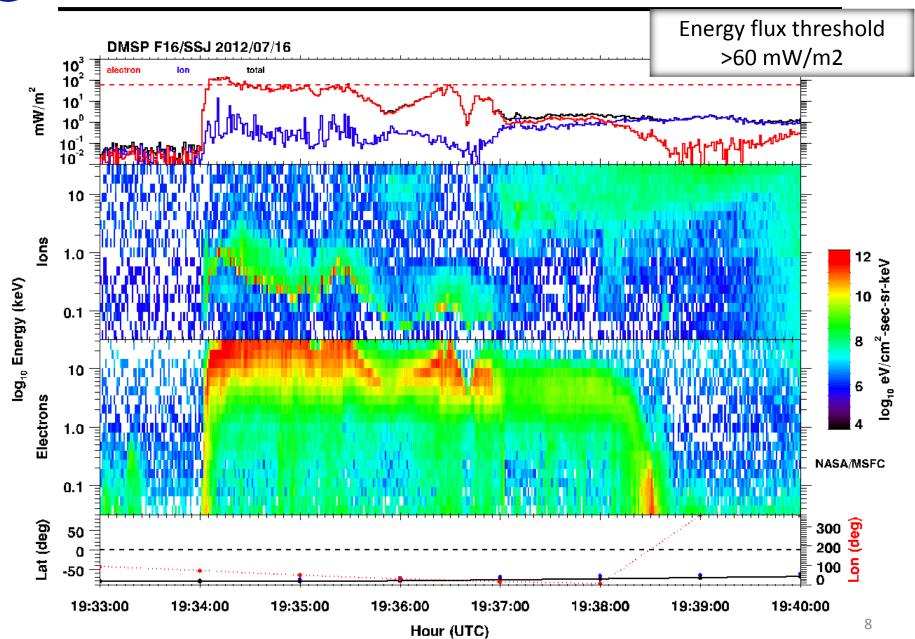


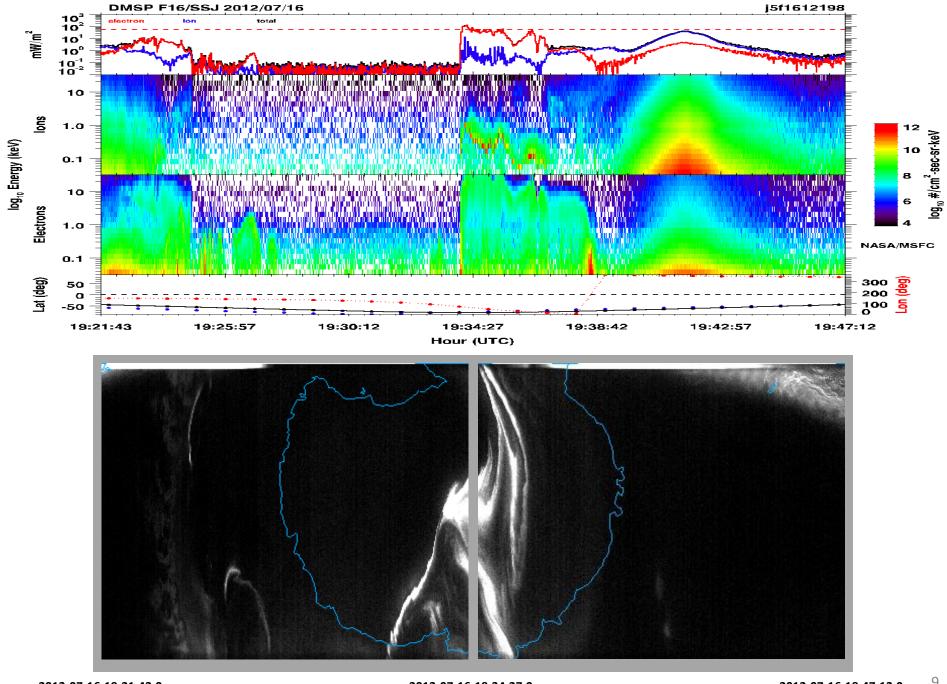
DMSP F16: -1000 V Charging Event

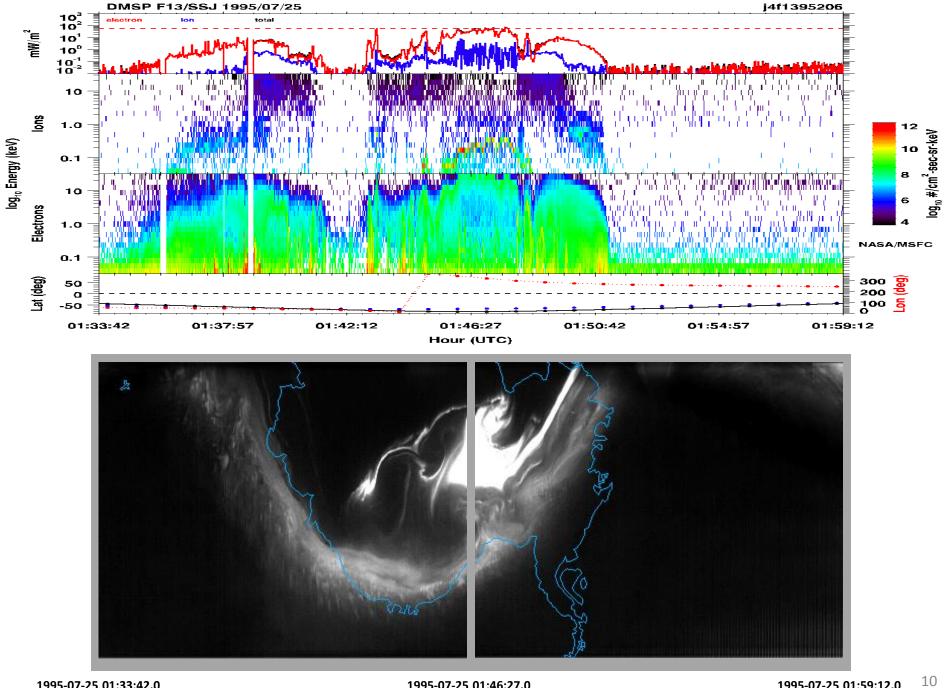




Energy Flux

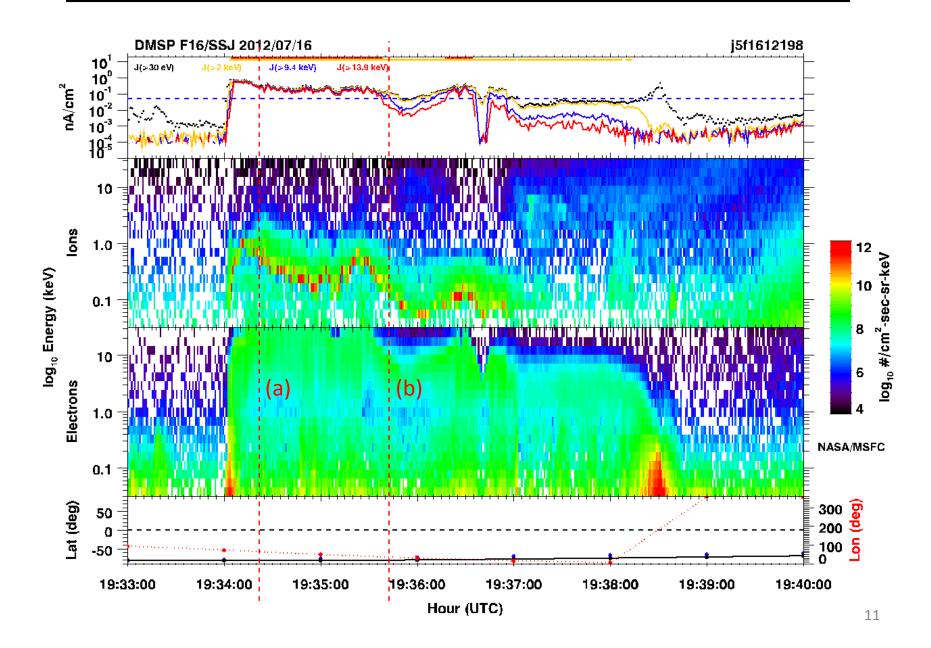






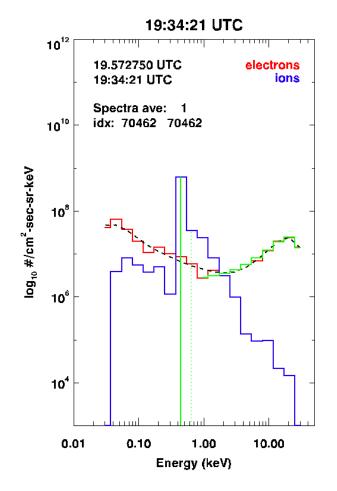


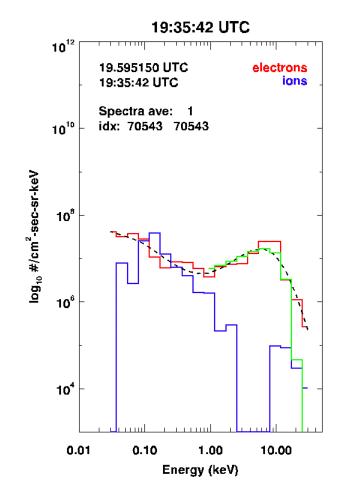
Individual Spectra





Individual Spectra





(a) (b)



Fontheim Distribution

Ambient background

n=1.0e10 1/m3

Te=0.2 eV

Maxwellian

Jmax = 4.0e-6 A/m²

Te = 3.0e3 eV

Gaussian (beam)

Jgau =0.9e-4 A/m²

Egau = 10.0e3 eV beam energy

dgau = 4.0e3 eV beam width

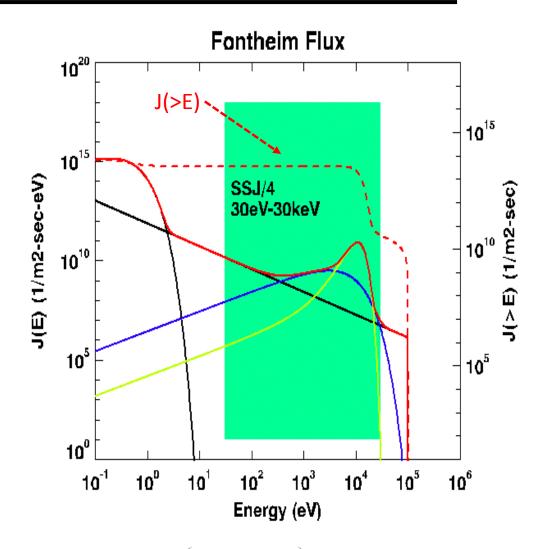
Power Law

Jpwr = 3.0e-7 A/m²

alpha = 1.15 exponent

E1=50.0 eV, first energy

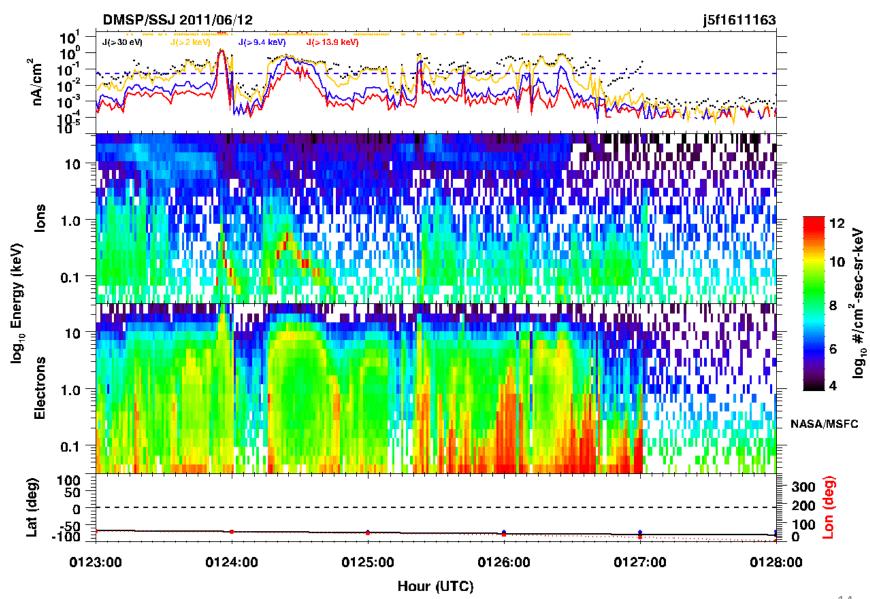
E2=1.0e5 eV, second energy



$$Flux\left(E\right) = \sqrt{\frac{e}{2\pi\theta m_{_{e}}}} \frac{E}{\theta} n \exp\left(-\frac{E}{\theta}\right) + \pi \zeta_{max} E \exp\left(-\frac{E}{\theta_{max}}\right) + \pi \zeta_{gauss} E \exp\left(-\left(\frac{E_{gauss} - E}{\Delta}\right)^{2}\right) + \pi \zeta_{power} E^{-\alpha}$$



Inverted V, Broadband Aurora

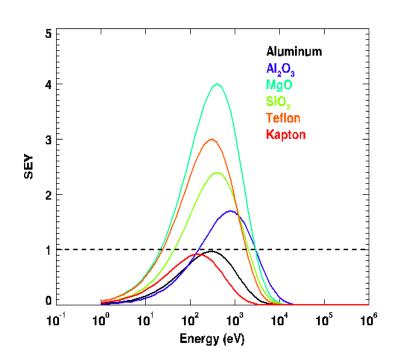




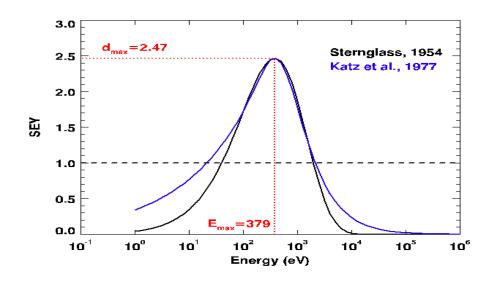
Secondary Electron Yields

Charging is suppressed when SEY > 1

$$\frac{dQ}{dt} = \sum_{k} I_{k} = +I_{i} - I_{e} + I_{se} + I_{ph,e}$$
$$= +I_{i} - I_{e} (1 - \delta) + I_{ph,e}$$



 $\delta_{\rm m}$, $E_{\rm m}$ from Hasting and Garrett, 1996



Sternglass, 1954

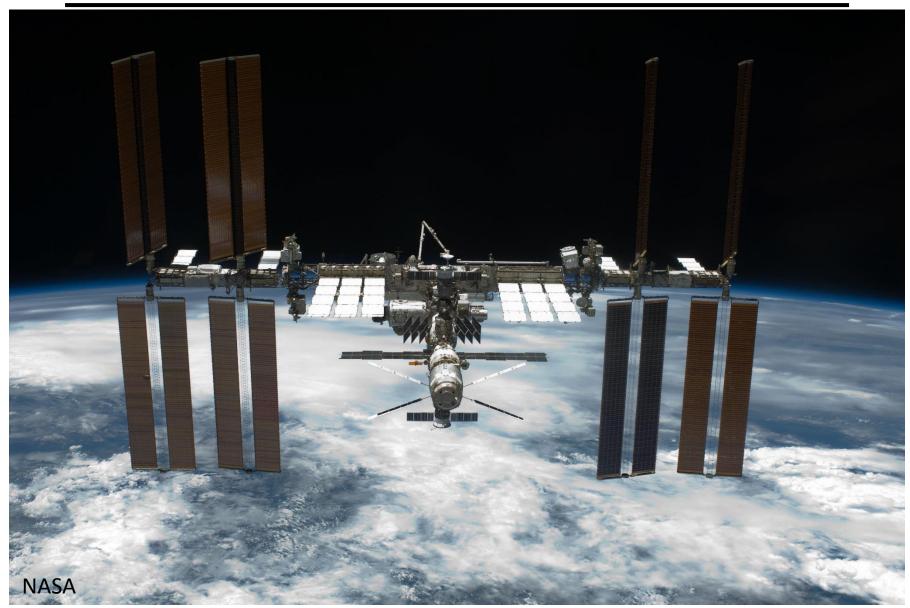
$$\delta_e(E, \theta) = \delta_{e, \text{max}} \frac{E}{E_{\text{max}}} \exp(2 - 2\sqrt{\frac{E}{E_{\text{max}}}}) \exp[2(1 - \cos \theta)]$$

Katz et al., 1977; Whipple, 1981

$$\delta_{e}(E,\theta) = \frac{1.114\delta_{e,max}}{\cos\theta} \left[\frac{E}{E_{max}} \right]^{0.35} \left\{ 1 - \exp\left[-2.28\cos\theta \left[\frac{E_{max}}{E} \right]^{1.35} \right] \right\}$$



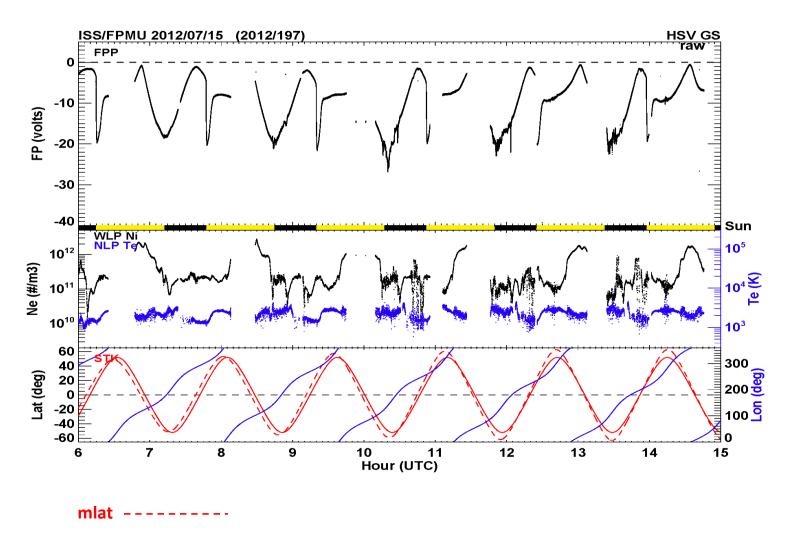
ISS Charging





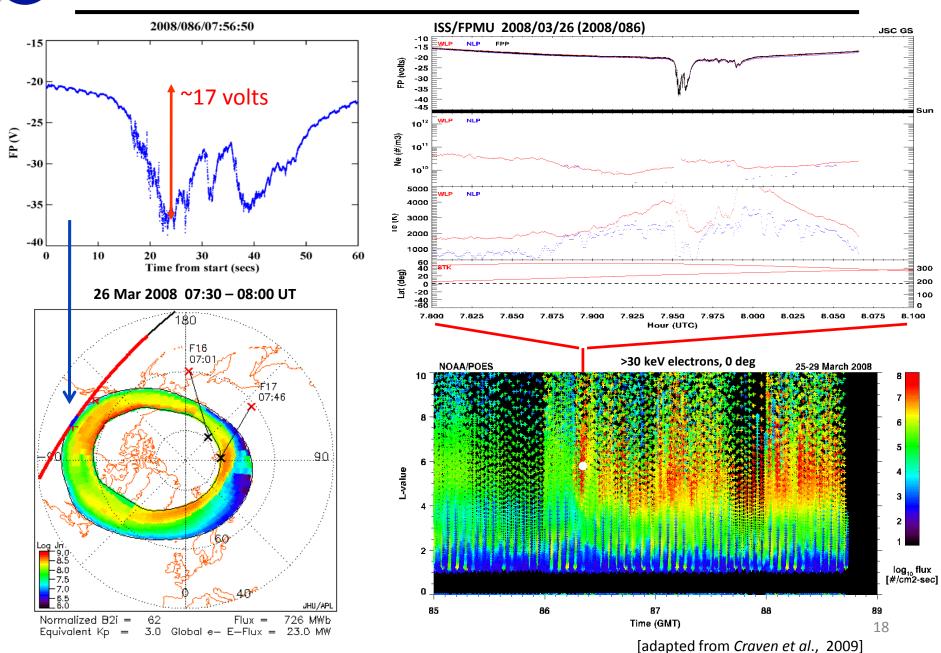
International Space Station: 15 July 2012

Potential variations due to (a) vxB.L (b) eclipse exit solar array (c) auroral charging



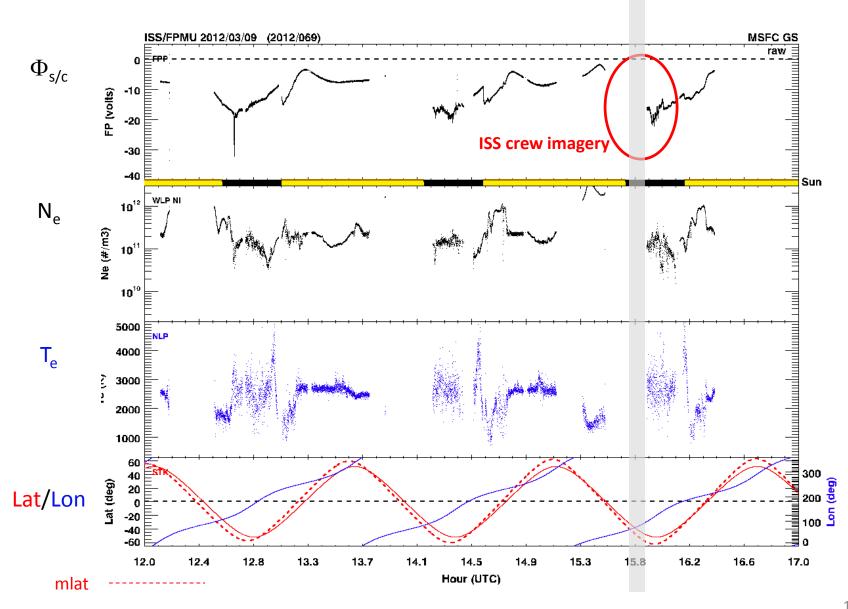


26 March 2008 -- Auroral Charging



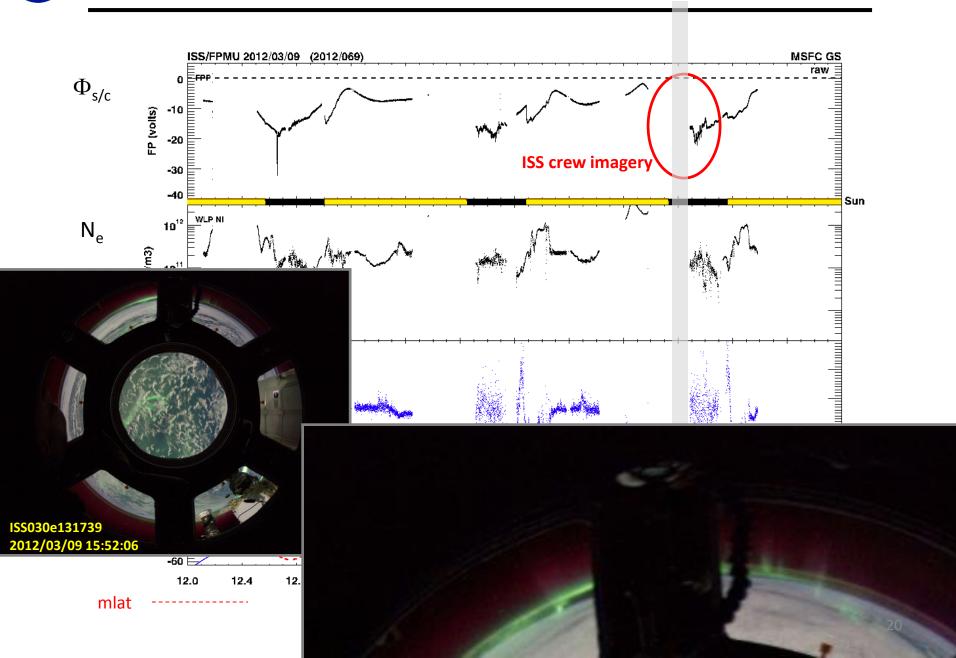


9 March 2012



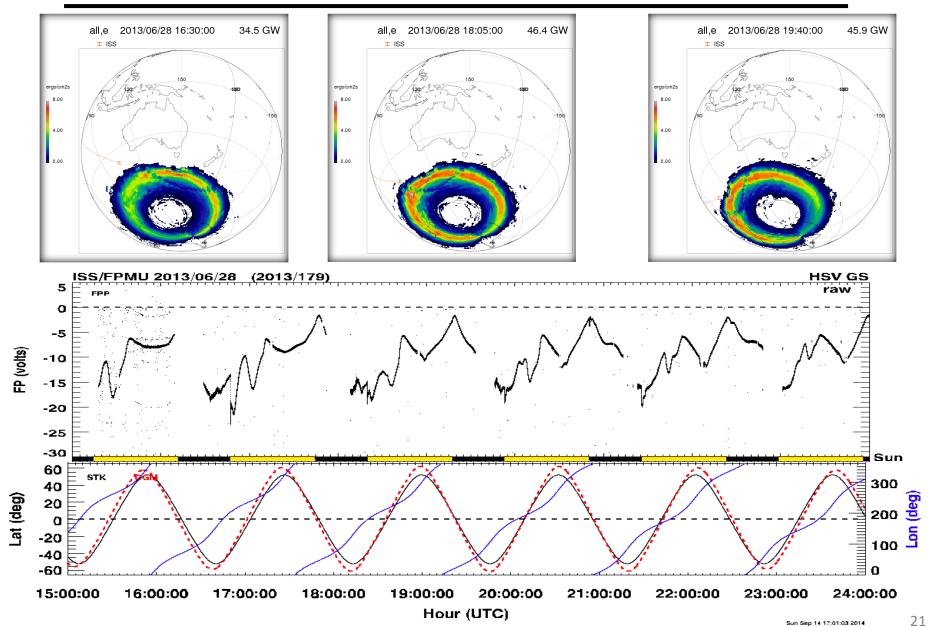


9 March 2012





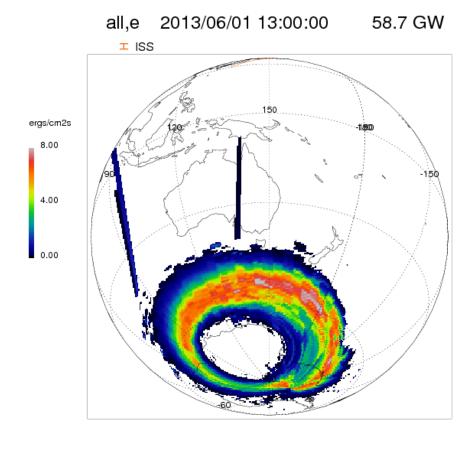
iSWA Ovation Prime, ISS Charging





Aurora Models

- NASA CCMC implementation of Ovation Prime is a good example of an auroral model providing total energy flux
- Total ions, electrons, and ions+electrons energy flux to 8 erg/cm²-s (=mW/m²)



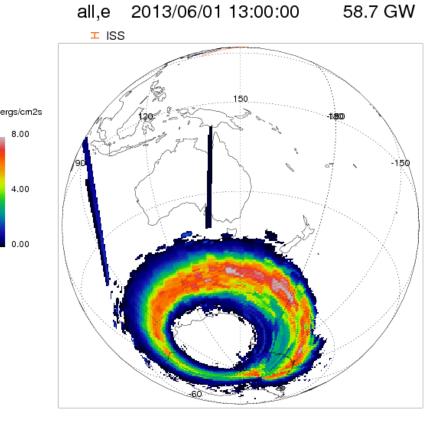
NASA CCMC

 $J \ge 8 \text{ ergs/cm}^2\text{-s}$



Aurora Models

- NASA CCMC implementation of Ovation Prime is a good example of an auroral model providing total energy flux
- Total ions, electrons, and ions+electrons energy flux to 8 erg/cm²-s (=mW/m²)
- Increase the energy flux coverage to include 10's to 100's ergs/cm2-s to consider auroral charging regime
- Energy flux for J_F(≥10 keV) erg/cm²-s



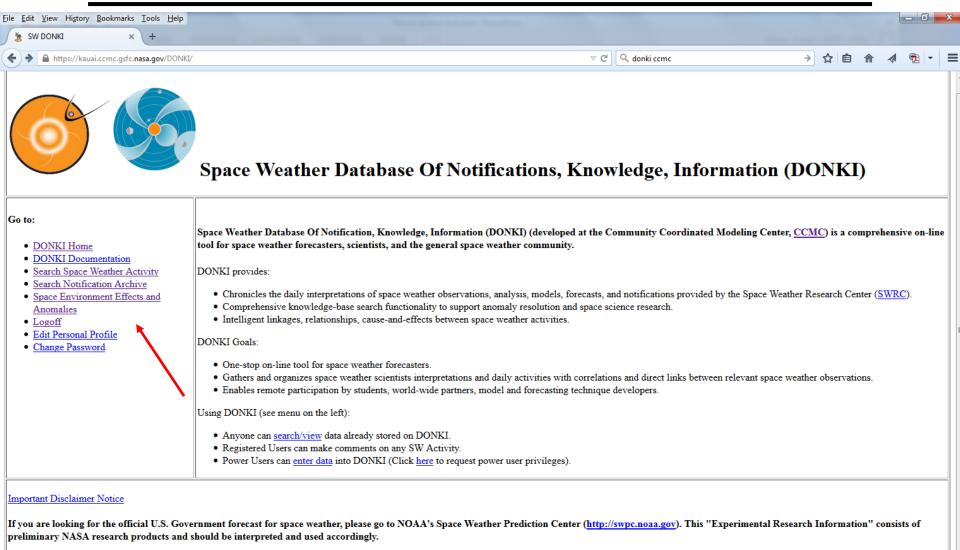
NASA CCMC

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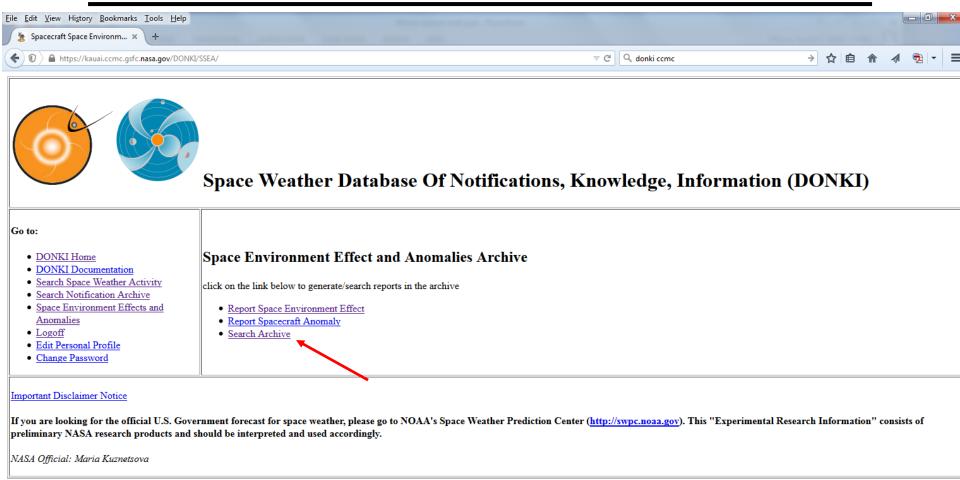
NASA Official: Maria Kuznetsova

CCMC DONKI





Space Environment Effect and Anomalies Archive



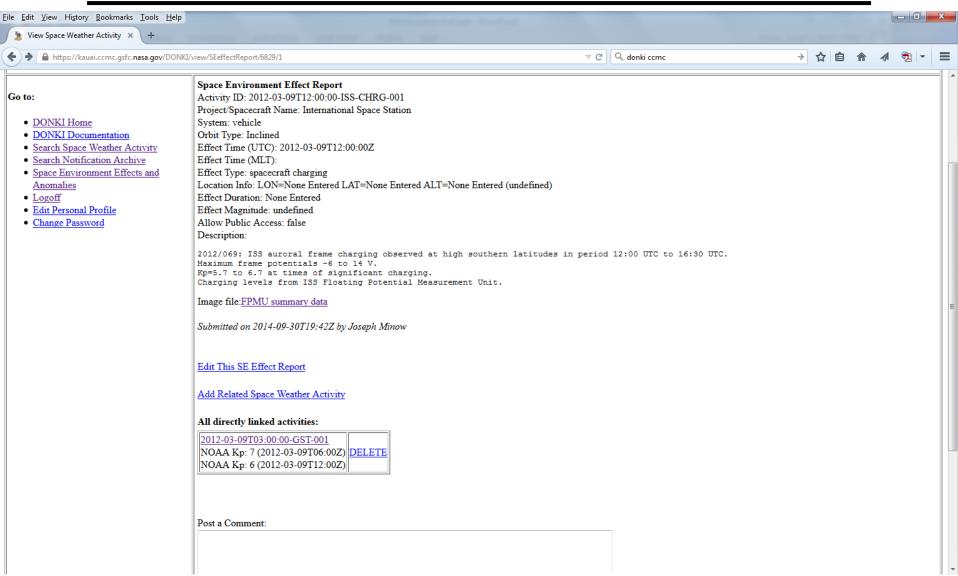


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Space Environment Effect Report





Space Environment Effect Report

